Potential Viability of a Fast-Acting Micro-Solenoid Valve for Pulsed Detonation Fuel Injection

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The potential for fast-acting solenoid valves to meet the demands of pulsed detonation fuel injection and other high-frequency devices is presented. The micro-valve was found to performance well above the manufacturer's rated frequency under no-load conditions, indicating the ability of spring-return micro-valves to function reliably at high frequencies. An examination of the electromagnetic force computed from the experimental displacement response demonstrates that flux saturation occurs early in the rise time. Opening and closing phases are examined and compared under ambient pressure conditions.

Nomenclature

A_x	cross sectional variable air gap normal to
	magnetic flux path (m ²)
C_d	damping coefficient (kg/s)
F_e	electromagnetic force of attraction (N)
F_{g}	gas pressure force exerted on the core in the
	x direction (N)
i_e	excitation current (A)
i_m	magnetization current (A)
k	spring constant (kg/s ²)
L_l	flux leakage inductance (H)
$L(\phi, x)$	non-linear coil inductance (H)
m	mass of movable valve core and spring
	assembly (kg)
n	integer number of successive (positive) peaks
	(Fig. 3)
N	number of coil turns
R_c	variable core loss resistance (Ω)
r_c	actual coil resistance (Ω)
T	period of oscillation (s)
V_s	supply voltage (V)
V_{ϕ}	flux leakage voltage (V)
W	energy stored in coil (J)
X	variable air gap length in the direction of the
	magnetic flux path (m)
\Re	total reluctance (A/Wb)
\Re_{m}	core reluctance (A/Wb)
$\delta^{^{ ext{}}}$	logarithmic decrement
μ_o	permeability of free space $(4\pi \times 10^{-7})$
, ~	Wb/A·m)
	,

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ϕ	magnetic flux (Wb)
ω_{d}	damped natural frequency (rad/s)
$\omega_{\rm n}$	undamped natural frequency (rad/s)
ζ	damping ratio

Introduction

 $\mathcal{T}^{\text{HERE}}$ is consensus, based on systems studies and from numerical simulations, that pulsed detonations on the order of at least 100 Hz are needed for aerospace propulsion systems. While there are various valveless concepts, they remain on the drawing boards. Practical demonstrators of pulse detonation technology tend to use mechanical, valve assemblies^{1,2} or solenoid valves for delivering the fuel, oxidizer and purge air. Most of these valves operate at frequencies lower than the theoretical requirements, partly due to limitations in valve perform-However, as cycling frequencies increase, less time is available to inject fuel. In many cases of gaseous fuel injection, the valve is already choked. Thus the injection orifice must be enlarged in order to deliver the required mass flow in the shorter available time and the fuel must be injected at higher pressure. Unfortunately, this equates to a slower response for typical rotary-type valves and hinders the system's ability to deliver the appropriate gases under stringent timing conditions. Further, in terms of overall engine design, weight and maintenance considerations tend not to favor these mechanical systems.

The possibility of utilizing high-frequency solenoid valves, with the advantages of electronic control, appears to be a viable solution for overcoming the limitations inherent in mechanical valves. These valves, with life spans of hundreds of millions of cycles, may be viable for missile and rocket applications. Solenoid valves are also made of materials compatible with the fluids used in pulse detonation engines. A search of the open literature reveals that solenoid valves used in pulse detonation ap-

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plications operate at low frequencies, typically 10-20 Hz.

To operate solenoid valves at the high frequencies of interest, valve behavior under high-frequency excitation must be understood. For example, a typical assumption of the fuel injection period is 30 percent of the cycle time.³ For 100 Hz operation, a satisfactory injection time might be 3 ms. Allowing a 10 percent deviation from an instantaneous response, an acceptable valve response would then be 0.3 ms.

Solenoid valve manufacturers usually do not provide information on response time or frequency response. Further, it is only recently that the dynamic behavior of solenoid valves has received comprehensive treatment, and few definitive works have been published.⁴⁻⁷ Hong⁶ developed a robust analytical model which can be adapted to most solenoid valve geometries. The model included the effects of core losses and flux leakage However, experimental data and some parameter estimation are still necessary to completely characterize a valve's response. Losses are best determined by experimentally measuring the solenoid force directly.^{5,6} Performance, however, can be examined through the relatively simple measurement of the displacement response. Such a method is presented in the following sections for a micro-solenoid valve and particularly examines performance above the maximum rated frequency at design voltages. The results point positively toward the potential that a scaled-up version of this design could be capable of meeting the requirements of pulse detonation fuel injection and other high-speed applications.

Valve Requirements

For this study, four primary characteristics were considered in the valve selection process: (1) operating frequency, (2) response time, (3) service life and (4) valve architecture. The first two are interdependent since valves rated for high-frequency operation typically possess a fast response. Additional criteria included seals appropriate for use with hydrogen, gaseous hydrocarbons and oxygen, high working pressures (approximately 100 psig) and elevated temperatures.

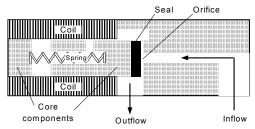


Figure 1. Cross-section of idealized solenoid valve.

A commercially available solenoid valve that partially fulfills the requirements stated above is one in the

Numatech® TM series, which has a maximum operating frequency of 50 Hz, as quoted by the manufacturer. Other pertinent information from the manufacturer includes a response time of 3 ms for energizing and 5 ms for de-energizing. Power consumption is 1.2 W (or 1.35 W, with LED indicator). This valve is a direct solenoid actuated, spring return, manifold mount, 3-way design available with Viton or Buna-N seals. The valve has a service life in excess of 200 million cycles and can operate at up to 690 kPa (100 psig). Limitations include a limited working temperature range of 0–50 °C, limited number of gaseous media and its small size. However, this valve possesses useful characteristics in that it can be dismantled and it can be easily modeled.

Depending on the configuration, the body is roughly $22 \times 27 \times 10$ mm. Fluid enters and exits the valve via three rectangular ports located side by side on the bottom of the valve, one inlet and two outlets. Each of the opening is 1 mm \times 3 mm square. Fluid, however, is injected into either of the two valve chambers through a 0.5 mm diameter circular cross-section orifice. Since the valve is a three-way design, there are two of these orifices.

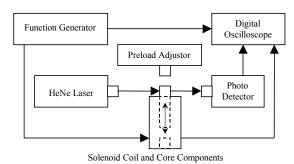


Figure 2. Schematic of experimental setup.

The valve design illustrated in Fig. 1 is particularly amenable to high-frequency applications and is comparatively easy to model. The movable portion of the design's split ferromagnetic core can be modeled as a simple mass-spring-damper system. A seal inserted in the end of the core blocks the injection orifice in the closed position and, when opened, allows the gas to expand rapidly into the chamber. A small vent through the core into the spring chamber serves as a mechanism to equalize the gas pressure as the valve chamber is filled. It is particularly advantageous since, as soon as the valve begins to open, the full orifice is available for injection. In rotary valves, on the other hand, the orifice opens and closes in phase, which may pose problems in controlling the flow.

Experimental Methodology

In order to observe the valve response independent of the valve casing, the solenoid coil and core components were separated from the valve assembly and secured in a brace to the original pre-loaded configuration. The solenoid coil and core components were tested using the setup illustrated schematically in Fig. 2.

A Metrologic (Blackwood, New Jersey) ML800 0.8 mW helium-neon laser was used to pass a confined beam of light perpendicular to the exposed movable core, the diameter of the beam being equivalent to the diameter of the core. A Thorlabs (Newton, New Jersey) Model DET200 high-speed photo detector (rise time of 1 ns, spectral range of 200 - 1100 nm) mounted on the opposite side measured the displacement response of the voltage-excited valve as a function of the time-varying light intensity. A BK Precision Model 3022 function generator supplied a square wave input signal of varying frequencies at approximately the manufacturer-rated input of 12 V. The input voltage, valve voltage, and photodetector output were recorded simultaneously by a Hewlett-Packard Model 5442A digital storage oscilloscope. A sampling rate of 25 kSa/s/channel sufficiently resolved the response in the opening and closing phases at all tested frequencies. A maximum sample length of 16386 data points was collected during each test, corresponding to a sample duration of 0.655 s. The oscilloscope trigger was set to initiate data collection five seconds into each test. The delay in data collection was chosen to allow the system sufficient time to stabilize.

Modeling the Dynamic Response

Most manifold mounted micro-valves are seated in the horizontal position, and (considering their small core masses) it is reasonable to neglect gravitational effects. Under this assumption the system dynamics can be represented by

$$m\ddot{x} + C_d \dot{x} + F_s = F_e + F_g \tag{1}$$

Unlike a free-response system, the motion of the core is constrained, and Eq. (1) is valid only over the allowed displacement range as determined by the valve geometry.

Spring Constant

At the full-open position, the spring is not completely compressed, but further displacement is limited by the fixed pole segment of the core. Assuming a linear relationship over the 0.5 mm displacement range, the spring force can be written as follows:

$$F_s = kx + 0.185 (2)$$

with a pre-load force of 0.185~N. Based on the manufacturer's data and using Eq. (2), the spring constant is 370~N/m. This value can be substantiated once the damping coefficient is determined as described in the following paragraphs.

Mass and Damping Coefficient

The combined mass of the movable core and spring were measured to be 0.605 ± 0.002 g using an Acculab V digital balance. The critical damping coefficient is given by

$$C_{d,crit} = 2\sqrt{km} = 0.946 \text{ kg/s}$$
 (3)

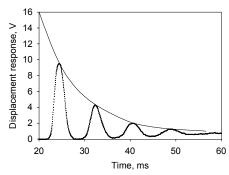


Figure 3. Undamped free response of valve core.

The logarithmic decrement describes the decay envelope of a free oscillating system. It is particularly amenable to the analysis of the damping ratio for simple mass/spring systems.

The free response to an initial displacement can be measured as a function of time by the method presented above. To do this, the spring is fixed to the core and the entire assembly is secured so that it is free to oscillate while partially constrained by the solenoid sleeve (see Fig. 1). An example of the data is shown in Fig. 3.

While the actual oscillation is sinusoidal, the negative portion of the signal shown in Fig. 3 is clipped since the valve core blocks the light beyond a 0.5 mm displacement. This is not an issue so long as the positive peaks are clearly defined and the true response is not constrained. The logarithmic decrement is then given by⁸

$$\delta = \frac{1}{n} \ln \left[\frac{x(t)}{x(t+nT)} \right] = 0.6128$$
 (4)

using the data of Fig. 3. As the number of peaks counted increases, the accuracy of the method increases as well. The damping ratio for this system can be expressed in terms of the logarithmic decrement to yield

$$\zeta = \delta / \sqrt{4\pi^2 + \delta^2} = 0.0971$$
 (5)

while the damped natural frequency is 785.4 rad/s. By definition, the natural frequency is

$$\omega_n = \omega_d / \sqrt{1 - \zeta^2} = \sqrt{k/m} = 789 \text{ rad/s}$$
 (6)

The spring constant was determined experimentally to be 377 kg/s^2 . These experimental results for the natural

frequency and spring constant are within 2 percent of that supplied by the manufacturer. The damping coefficient is given by

$$C_d = 2\zeta \sqrt{km} = \zeta C_{d,crit} = 0.0927 \text{ kg/s}$$
 (7)

Gas Pressure Force

Micrometer measurements of the orifice diameter were Similarly, micrometer measurements of the maximum core displacement—measured from the preloaded to full-open position—were also 0.5 mm. At the maximum rated pressure of 100 psig, the gas pressure force through the orifice is approximately 0.136 N. The manufacturer specified the spring preload force at 0.185 N—sufficient to overcome this force—while the final load in the full-open position was stated at 0.370 N. The small maximum displacement would do little to dissipate the pressure force exerted against the core tip, thus it should be sufficient in subsequent analyses to assume the gas pressure force to be constant. All experiments were conducted without flow. Thus the gas pressure term has been excluded from Eq. (1) in the analysis. The results of the previous discussion can now be used to rewrite Eq. (1) in a form that is representative of the actual experimental conditions and constraints:

$$6.05 \times 10^{-4} \ddot{x} + 9.27 \times 10^{-2} \dot{x} + 376.75 x = F_e - 0.185$$
 (8)

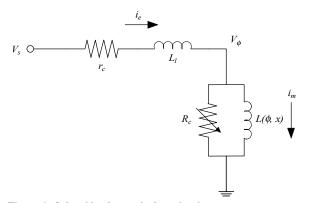


Figure 4. Solenoid valve equivalent circuit.

Force Characteristics of a Non-linear Solenoid

It is useful at this point to discuss the relationship between magnetic flux and electromagnetic force so as to more fully comprehend the nature of solenoid optimization. Figure 4 shows an equivalent circuit for a solenoid valve that helps to explain the nonlinear behavior of the valve's voltage and displacement response. The circuit elements include the actual coil resistance r_c , a variable flux leakage inductance L_l (which is due to flux lines leaking through the air gaps and is responsible for shifting the flux versus current curve to the left or right), a non-linear coil inductance $L(\phi, x)$, and a variable coreloss resistance R_c (which results from eddy current losses and determines the shape of the hysteresis loop about the steady-state magnetization curve—the larger its value, the wider the hysteresis loop).

For a typical solenoid, there are two sources of core loss: eddy currents and hysteresis. Eddy currents circulating in the core material are the direct result of core voltages induced by the changing magnetic field. Hysteresis is the result of the energy dissipating properties of the core itself, acting as a variable resistance. The losses result in a difference between the exciting and magnetizing currents, i_e and i_m , see equivalent circuit in Fig. 4. In other words, were there no losses, these values would be identical.

When current flows through a simple coil, magnetic flux is generated. The product of the magnetizing current and the number of coil turns is often referred to as the magnetomotive force (though not in the dimensional sense) and can similarly be expressed as the product of magnetic flux and total reluctance, $i_m = \frac{\phi \cdot \Re}{N}$

$$\dot{i}_m = \frac{\phi \cdot \Re}{N} \tag{9}$$

The energy delivered to the coil can be expressed as the integral of power in terms of total reluctance and flux¹⁰

$$W = \int_0^{\phi} Ni_m \cdot d\phi = \int_0^{\phi} \Re \phi \cdot d\phi \tag{10}$$

The electromagnetic force is expressed as the negative of the partial differential of the stored energy with respect to the variable air gap distance, the negative sign indicating the direction of the force, 10

$$F_e = -\frac{\partial w}{\partial x} = -\frac{\partial}{\partial x} \left(\int_0^{\phi} \Re \phi \cdot d\phi \right)$$
 (11)

using Eq. (10).

The total reluctance is equal to the sum of the component reluctances that include those from: the fixed air gap, the variable air gap, and the core material (see Appendix). 11,12 Of these three terms, only the variable air gap contains the displacement x so that Eq. (11) simplifies to

$$F_e = -\frac{\partial}{\partial x} \left(\int_0^{\phi} \frac{x\phi}{\mu_o A_x} d\phi \right) = -\frac{\phi^2}{2\mu_o A_x}$$
 (12)

The results clearly show the quadratic relationship between magnetic flux and electromagnetic force. At some point, the magnetic flux saturates. This point corresponds with the point of maximum force or minimum core loss. It is desirable for flux saturation to occur as quickly as possible thus maximizing the electromagnetic force and optimizing the opening phase of the solenoid valve.

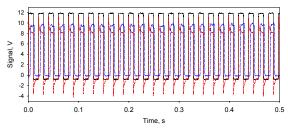
Experimental Results

The valve was tested at frequencies beginning with the manufacturer's stated maximum of 50 Hz up to 110 Hz in 10 Hz intervals and then again at 150 Hz. At a sampling rate of 25 kSa/s, the cycle accuracy for all tests was 40 μs . This corresponds to an accuracy of ± 0.2 –0.6 percent of the measured frequency at 50–150 Hz. The sample length allowed approximately 33 consecutive cycles to be observed at 50 Hz and approximately 98 cycles at 150 Hz. Thirty cycles from each data set were used to construct averaged response plots.

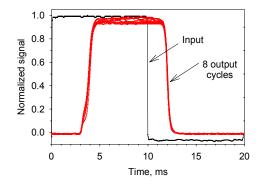
A set of sample results is shown in Fig. 5 for a 50 Hz duty cycle with the displacement response scaled to the input signal. Figure 5a shows a sample of the data. The excellent repeatability is demonstrated in Fig. 5b by overlaying eight consecutive cycles. The average of thirty consecutive cycles is shown enlarged in Fig. 5c. In Figs. 5b and c, the input and solenoid output voltages are

normalized by the averaged, measured input value while the photodetector output is normalized by the maximum photodetector output.

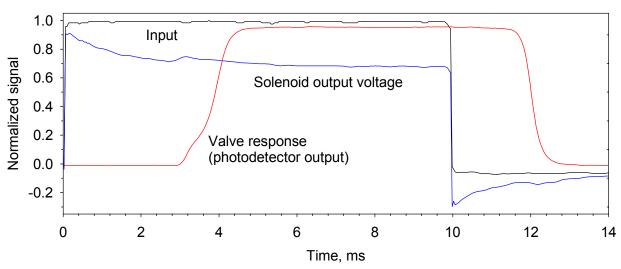
The 12 V, square-wave excitation shows a slight negative offset. The solenoid output immediately follows the excitation. Motion of the core begins when the magnetic field strength is sufficient to overcome the spring force and draw the core into the field. Initially the boss end extends slightly beyond the field influence and eddy currents are low. However, as the core moves more fully into the field, eddy currents increase resulting in a corresponding temporary decrease in the magnetizing current. As this occurs, the solenoid force is weakened. resulting in a temporary slowdown in core displacement as seen by the slight reduction in the slope of the displacement curve. Fig. 5c. Once flux saturation occurs. the core loss resistance stabilizes, the flux leakage inductance decreases which increases the magnetizing current, and, consequently, the electromagnetic force and rate of



a. Data.



 $b. \ \ Repeatability\ check-eight\ consecutive\ cycles.$



c. Average of eight cycles.

Figure 5. Input and outputs at 50 Hz excitation.

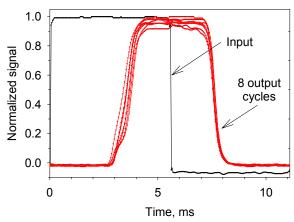


Figure 6. Repeatability check at 90 Hz- eight consecutive cycles.

displacement. The rise in voltage observed just prior to the opening displacement phase is the result of an induced voltage created by the changing magnetic field. While still present, these effects are less pronounced during the closing phase as the magnetic field collapses. The phenomena described above are observed in all data samples and at all frequencies tested. For example, Figs. 6 and 7 show the normalized output of eight cycles at 90 and 100 Hz operation. Some deterioration in opening phase repeatability is evident at 100 Hz, and significant cycle deterioration is evident at 150 Hz, the highest frequency tested (Fig. 8).

The opening and closing response times of the valve at 50–110 Hz are shown in Fig. 9. It is important to keep in mind that the geometry of the valve tested (Fig. 1) possesses a significant advantage over other valve types, specifically rotary types—this being that there is no partial open or closed position. The slightest motion of the core breaks the seal, allowing the immediate inflow of gas. While the data illustrates a finite time to fully open or close the valve, it is expected that the mass flow will

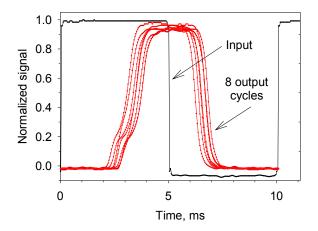


Figure 7. Repeatability check at 100 Hz- eight consecutive cycles.

closely match the input signal characteristics rather than the core response, being primarily governed by gasdynamic considerations. If this is indeed the case, whether or not the valve fully opens will not be as critical as the meticulous control of the cycle time as a means of delivering a precise gaseous quantity. Thus—considering the desired 0.3 ms response time as stated previously—it is not unreasonable to expect that maximum gas flow might be achieved when the valve is only partially retracted. Figure 9 shows that the rise and fall times do not vary with frequency but remain virtually constant, while the increasing phase lag tends to diminish the full open time. These observations confirmed unpublished analysis.

Based on the data in Fig. 9, the combined averaged time from initial excitation to full displacement is approximately 8.6 ms, implying that, at any frequency above 116 Hz, the input signal will be terminated prior to the valve being completely open. Thus, in order for valid tests to be conducted above this frequency, the duty cycle of the input signal would have to be increased. For example, this can be seen by the obvious cycle failures in

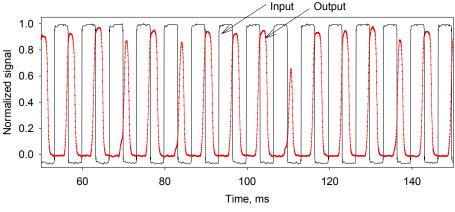


Figure 8. 150 Hz.

Fig. 8.

Figure 9 illustrates an unexpected result. Over the range of tested frequencies, the valve's dynamic performance at the recommended input voltage appears optimized at 70-90 Hz, or 140 to 180 percent above the manufacturer's stated maximum frequency (without consideration given to valve longevity). When combined with the results of repeatability, the range of 70-80 Hz is clearly optimal. These results may be due to hysteresis effects or differences in the loss mechanisms caused by the higher frequency input so long as the maximum frequency for the input signal duty cycle is not exceeded. A more detailed examination of the phenomena is required before definite conclusions can be drawn. Nevertheless, the results point encouragingly toward the capability of simple spring-return solenoid devices to function at the frequency levels demanded by pulse detonation applications, consistently and reliably.

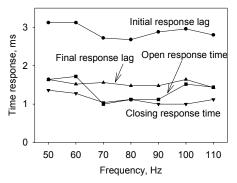


Figure 9. Response characteristics.

Conclusions

A miniature solenoid valve was tested under no-load conditions. The architecture of this particular valve type facilitated the understanding of valve operation at high frequency. It was found that the valve opening lagged by a constant time delay for a range of frequencies from 50–110 Hz when excited at 12 V. The motion of the solenoid core could be understood in terms of the interaction between the electromagnetic and spring forces. The magnetizing current lags the exciting current creating a phase delay at all frequencies. When the cycle frequency increases, however, the cycle-to-cycle repeatability deteriorates and the solenoid's behavior became erratic at 150 Hz.

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Appendix: Development of Solenoid Reluctance

Theoretically, if the material permeability can be assumed constant, the reluctance is equal to the length of the path in the direction of the magnetic flux divided by the product of the permeability and the effective area normal to the path. For most solenoids, three reluctances must be considered: the fixed air gap reluctance, the variable core air gap reluctance, and the core material reluctance. The first two involve air whose permeability can be assumed constant, and the standard equation can be used. The reluctance of the fixed air gap is then

$$\mathfrak{R}_g = \frac{g_f}{\mu_o A_o} \tag{A1}$$

where A_g and g_f are the fixed air gap cross-sectional area and the fixed air gap length, respectively, both in the direction of the magnetic flux path. The reluctance of the variable core air gap is

$$\Re_x = \frac{x}{\mu_o A_x} \tag{A2}$$

where A_x and x are the variable air gap cross-sectional area and the variable air gap length, respectively, both in the direction of the magnetic flux path. The core reluctance can be expressed as

$$\mathfrak{R}_{m} = \frac{f_{2}}{f_{1} - \phi} \tag{A3}$$

where f_1 and f_2 are constants related to the magnetization characteristics of ferromagnetic materials.⁶

From the definition of the magnetomotive force, the magnetizing current in a simple coil can be expressed in terms of the product of flux and total reluctance divided by the number of coil turns,

$$i_m = \Re_g + \Re_x + \Re_m = \frac{\phi}{N} \cdot \left(\frac{g_f}{\mu_o A_g} + \frac{x}{\mu_o A_x} + \frac{f_2}{f_1 - \phi} \right)$$
 (A4)

which is equivalent to Eq. (9) in the main text.